

A normal stress subgrid-scale eddy viscosity model in large eddy simulation

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1. Motivation and objectives

The Smagorinsky subgrid-scale eddy viscosity model (SGS-EVM) is commonly used in large eddy simulations (LES) to represent the effects of the unresolved scales on the resolved scales. This model is known to be limited because its constant must be optimized in different flows, and it must be modified with a damping function to account for near-wall effects. The recent dynamic model (Germano *et al.* 1991) is designed to overcome these limitations but is compositionally intensive as compared to the traditional SGS-EVM. In a recent study using direct numerical simulation data, Horiuti (1993) has shown that these drawbacks are due mainly to the use of an improper velocity scale in the SGS-EVM. He also proposed the use of the subgrid-scale normal stress as a new velocity scale that was inspired by a high-order anisotropic representation model (Horiuti 1990). The testing of Horiuti (1993), however, was conducted using DNS data from a low Reynolds number channel flow simulation. It was felt that further testing at higher Reynolds numbers and also using different flows (other than wall-bounded shear flows) were necessary steps needed to establish the validity of the new model. This is the primary motivation of the present study. The objective is to test the new model using DNS databases of high Reynolds number channel and fully developed turbulent mixing layer flows. The use of both channel (wall-bounded) and mixing layer flows is important for the development of accurate LES models because these two flows encompass many characteristic features of complex turbulent flows.

2. Accomplishments

The subgrid-scale stress tensor, τ_{ij} , that results from filtering the Navier-Stokes equations consists of three terms (Bardina 1983):

$$\tau_{ij} = L_{ij} + C_{ij} + R_{ij}, \quad (1)$$

$$L_{ij} = \overline{\overline{u_i u_j}} - \overline{u_i} \overline{u_j}, \quad C_{ij} = \overline{\overline{u_i u'_j} + u'_i \overline{u_j}}, \quad R_{ij} = \overline{u'_i u'_j}$$

where *overline* u_i denotes the filtered velocity component and $u'_i = u_i - \overline{u_i}$ denotes the SGS component of u_i . L_{ij} is the Leonard term, C_{ij} is the cross term, and R_{ij} is the SGS Reynolds stress. The indices $i = 1, 2, 3$ correspond to the directions

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x , y , and z , respectively, with x the streamwise ($u_1 = u$), y the “major-gradient” (wall-normal or cross-stream) ($u_2 = v$), and z the spanwise ($u_3 = w$) directions.

The Leonard term in eq. (1) is not modeled but is treated explicitly by applying the filter, while the other two terms (C_{ij} and R_{ij}) need to be modeled. A successful model for the cross term is a model suggested by Bardina (1983) where

$$C_{ij} = \overline{u'_i \overline{u}_j} + \overline{\overline{u}_i u'_j}$$

This model has been tested by Bardina (1983) for homogeneous flows and by Horiuti (1989) for the channel flow and was found to be a good model for the cross terms. This model will not be tested further in this work.

For the R_{ij} terms, the eddy viscosity model by Smagorinsky (Smagorinsky 1963):

$$R_{ij} \sim \frac{2}{3} \overline{E}_G \delta_{ij} - \nu_e \left(\frac{\partial \overline{u}_i}{\partial x_j} + \frac{\partial \overline{u}_j}{\partial x_i} \right), \quad (2)$$

$$\nu_e = (C_S \Delta)^2 \left[\frac{1}{2} S_{ij} S_{ij} \right]^{1/2}, \quad S_{ij} = \frac{\partial \overline{u}_i}{\partial x_j} + \frac{\partial \overline{u}_j}{\partial x_i},$$

and the Bardina model

$$R_{ij} \sim C(\overline{u}_i - \overline{\overline{u}}_i)(\overline{u}_j - \overline{\overline{u}}_j). \quad (3)$$

are two of several models which are used in LES computations. In these models, C_S and C are model constants, and $\overline{E}_G = \overline{u'_i u'_i} / 2$ and ν_e are, respectively, the SGS turbulent kinetic energy and SGS eddy viscosity coefficient. Δ is the characteristic SGS length scale whose value is defined as $(\Delta x \Delta y \Delta z)^{1/3}$; Δx , Δy , and Δz are the grid intervals in the x , y , and z directions, respectively. The Smagorinsky model is a “Prandtl-type” mixing length model that can be derived by starting with the eddy viscosity approximation to the subgrid-scale Reynolds stresses and assuming production and dissipation are in balance. In an eddy viscosity approximation, ν_e is written as the product of a characteristic time scale τ and a velocity scale $E^{1/2}$,

$$\nu_e = C_\nu \tau E \quad (4)$$

where C_ν is a model constant. τ is then expressed as (Horiuti 1993)

$$\tau = \frac{\overline{E}_G}{\epsilon}, \quad \epsilon = \nu \frac{\partial \overline{u}'_i}{\partial x_l} \frac{\partial \overline{u}'_i}{\partial x_l} = C_\epsilon \frac{\overline{E}_G^{3/2}}{\Delta}, \quad (5)$$

where ϵ is the dissipation rate of \overline{E}_G and C_ϵ is a model constant. The Smagorinsky model assumes that $E = \overline{E}_G$ in (4).

In the present study, we make use of the direct numerical simulation flow fields available at CTR to directly test the various approximations. The fields we consider are homogeneous in two-directions. To compute the large-eddy flow fields, we filter the DNS fields by applying a two-dimensional Gaussian filter in the $i = 1, 3$ directions. In the inhomogeneous direction ($i = 2$), a top-hat filter is applied to the

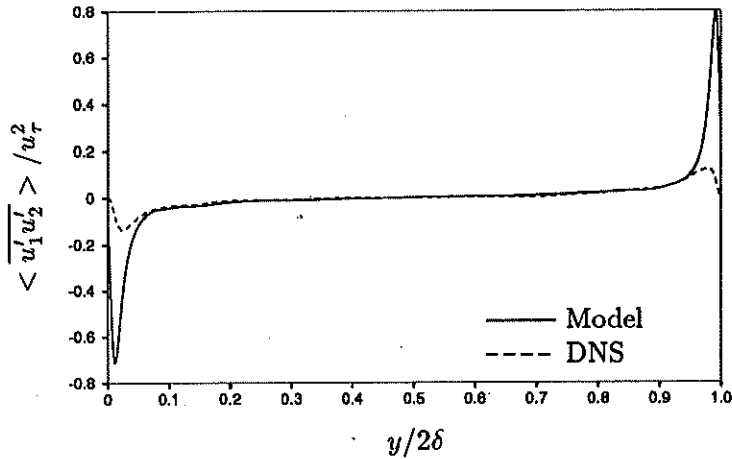


FIGURE 1. y -distribution of the SGS-Reynolds shear stress in turbulent channel at $Re_\tau = 790$. (Model with $E = \overline{E}_G/u_\tau^2$.)

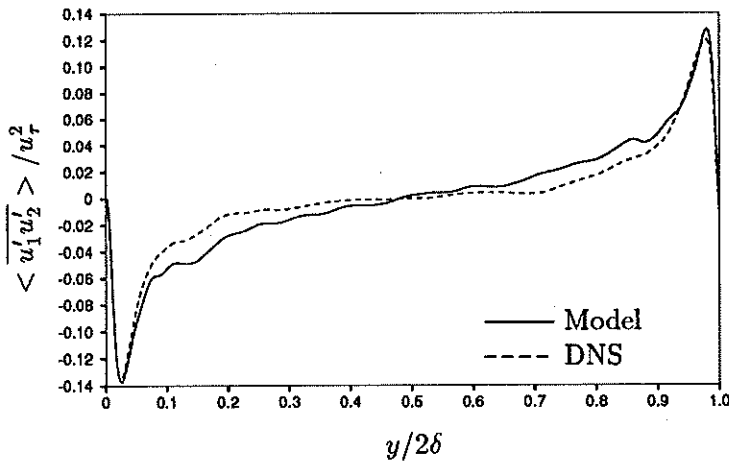


FIGURE 2. y -distribution of the SGS-Reynolds shear stress in turbulent channel at $Re_\tau = 790$. (Model with $E = \overline{u_2' u_2'} / u_\tau^2$.)

channel flow fields. No filter was applied in this direction ($i = 2$) to the mixing layer flow field. This is due to the fact that occasionally the doubly filtered ($\overline{\overline{\cdot}}$) grid-scale variables were larger than the singly filtered ones ($\overline{\cdot}$), owing to the inaccuracy of a top-hat filter in regions where grid spacing is coarse.

The DNS databases we used were the fully developed incompressible channel flows at Re_τ (Reynolds number based on the wall-friction velocity, u_τ , and the channel height, 2δ) = 360 (Kim *et al.* 1987) and 790 (Kim 1990), and the incompressible mixing layer at Re_θ (the Reynolds number based on the momentum thickness, δ_m ,

and the velocity difference, $\Delta U = 2400$ (Moser and Rogers 1992). We started with the low Reynolds number channel flow data as a confidence test. We found that the results obtained in this case are consistent with the previous work of Horiuti (1993), who used a different set of DNS data but at the same Reynolds number. The details of this testing are not shown in the present report.

The high Reynolds number channel flow field (with $256 \times 193 \times 192$ grid points) was filtered to $64 \times 97 \times 48$ grid points. The mixing layer flow field (with $512 \times 210 \times 192$ grid points) was filtered to $64 \times 210 \times 48$ grid points. These LES grid point numbers were chosen so that the turbulent kinetic energy retained in the SGS components is large. This is needed to make a fair assessment of the SGS models. SGS model evaluations were conducted by comparing the y -distribution of the mean values averaged in the $x - z$ plane (denoted by $\langle \rangle$), and also by comparing the y -distribution of the root-mean-square (rms) values of the exact terms with the model predictions. Only the y -distribution of the mean values are shown in the present report because the rms values were found to give similar results.

2.1 A proper eddy viscosity velocity scale

2.1.1 Channel flow

The y -distribution of the SGS Reynolds shear stress $\langle \overline{u'_1 u'_2} \rangle$ obtained with $E = \overline{E}_G$ and $C_\nu = 0.1$ in (4) is compared with the DNS data in Fig. 1. While the agreement between the model and the term is good in the central portion of the channel, the agreement deteriorates near the wall where the model predicts a very large peak compared to the actual data. This overprediction of the shear stress near the wall when \overline{E}_G is used for E in (4) implies that a damping function is needed to account for the presence of the wall. This near-wall overprediction of the stress is similar to the near-wall behavior of one-point closure models (see Rodi & Mansour 1991). This behavior of one-point closure models is attributed to the rapid reduction of the Reynolds shear stress (as the wall is approached) due to the preferential damping of the normal stress (Launder 1987, and Durbin 1992). Horiuti (1993) reasoned that the same wall damping effects should hold true for the SGS field. Indeed, when the SGS normal stress $\overline{u'_2 u'_2}$ is used for E (with $C_\nu = 0.23$, see Fig. 2), the model agrees well with SGS Reynolds shear stress near the wall without an additional damping function. The model is, however, less effective as compared to using the total energy in the core region of the channel. The main deficiency in the core region is attributed to excessive grid stretching in the y -direction because of the mapping used in conjunction with Chebyshev expansions. In an actual LES computation, finite differences with a more uniform grid are used in the y direction and, therefore, a more isotropic energy distribution can be expected in this case.

The effects of the anisotropic grid can be evidenced by the y -distribution profile of the 'flatness parameter' A (Lumley 1978) averaged in the $x - z$ plane. In this case A is defined as

$$A = [1 - \frac{9}{8} \{A_2 - A_3\}], A_2 = a_{ij} a_{ij}, A_3 = a_{ij} a_{jk} a_{ki}, \quad (6)$$

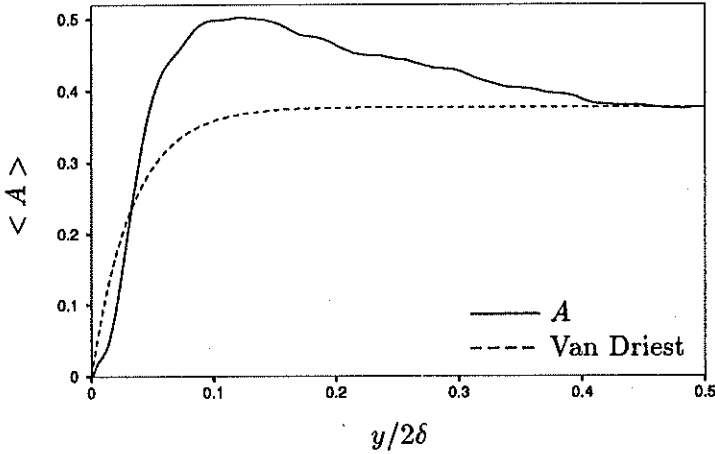


FIGURE 3. y -distribution of the flatness parameter A and the Van Driest function (channel at $Re_\tau = 790$).

$$a_{ij} = \{\overline{u'_i u'_j} - \frac{1}{3} \delta_{ij} \overline{u'_k u'_k}\} / k, k = \frac{1}{2} \overline{u'_k u'_k}$$

We find (see Fig. 3) that in the core region of the channel, $A \sim 0.35$, which is much smaller than the expected $A = 1$ when the small scale turbulence is isotropic. In the region around $y \sim 0.1$, A peaks around $A \sim 0.5$, and then gradually decreases to 0.35 at the channel center. The y -distribution of A for the unfiltered DNS data does not show this overshoot and is close to $A = 1$ around the centerline. The grid spacing in the central region of channel seems to be too coarse; therefore, a considerable anisotropy exists in the SGS turbulence fluctuations. In fact, when the SGS-EVM model with $E = \overline{u'_2 u'_2}$ in (4) was used in an actual LES channel flow calculations using a more uniform grid at high Reynolds number ($Re_\tau = 1280$) (Horiuti 1993), a good agreement with experimental data was found. The present comparisons for the high Re channel flow confirm the conclusions of Horiuti (1993) based on the low Re channel flow fields.

For the record, the y -distribution of the conventional Van Driest damping function $((1 - \exp(-y^+/26.0))$ (normalized with value of A at the channel center) is included in Fig. 3. It should be noted that the 'flatness parameter' A has a similar distribution across the channel as the Van Driest function, suggesting that A may be used as an alternative method to damp the eddy viscosity near the wall (Horiuti 1992).

2.1.2 Mixing layer

The y -distribution of $\overline{u'_1 u'_2}$ obtained using $E = \overline{E}_G$ ($C_\nu = 0.20$) and $E = \overline{u'_2 u'_2}$ ($C_\nu = 0.26$) in (4) are compared with the DNS data in Fig. 4 and 5, respectively. Both cases show a good agreement of the model with the DNS data, indicating that the two models are equivalent in this case. It should be noted that the optimized C_ν values obtained for the $\overline{u'_2 u'_2}$ model in the channel flow at lower Re (0.22), at high Re

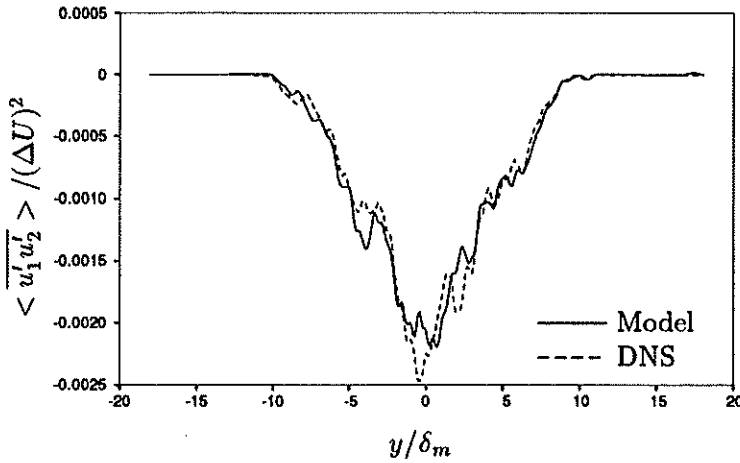


FIGURE 4. y -distribution of the SGS-Reynolds shear stress in turbulent mixing layer. (Model with $E = \overline{E}_G / (\Delta U)^2$.)

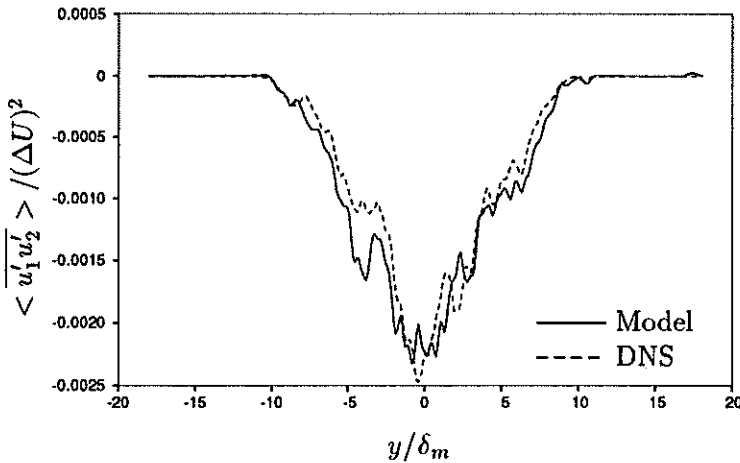


FIGURE 5. y -distribution of the SGS-Reynolds shear stress in turbulent mixing layer. (model with $E = \overline{u_2' u_2'} / (\Delta U)^2$)

(0.23), and in the mixing layer (0.26) are very close to each other. This implies that the model constant of the SGS normal stress model is rather universal independent of the type of flow, whereas the optimized C_ν values for the $E = \overline{E}_G$ model were 0.11, 0.10, 0.20, respectively. This is further indication of the potential strength of the normal-stress model. It is interesting to note that, particularly in the outer edge region of the mixing layer ($y \sim 10$ or ~ -10), the magnitude of normal component was the largest among the three components of the SGS turbulence fluctuations. A possible relationship of this phenomenon with the significant intermittency in these regions will be investigated in future work.

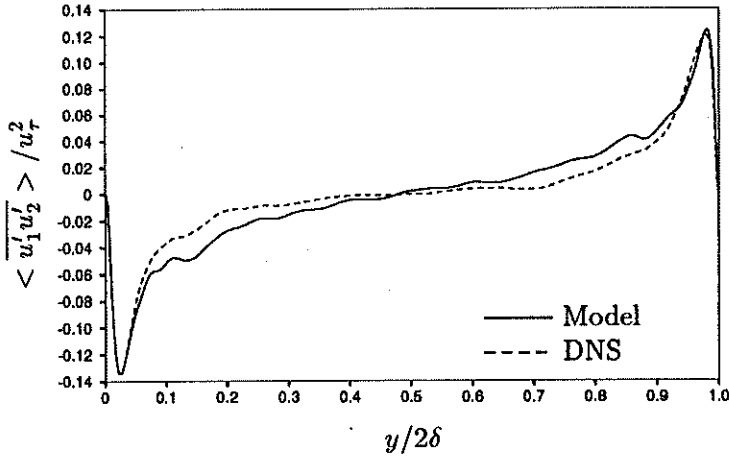


FIGURE 6. y -distribution of the SGS-Reynolds shear stress using Generalized normal stress model (Channel at $Re\tau = 790$)

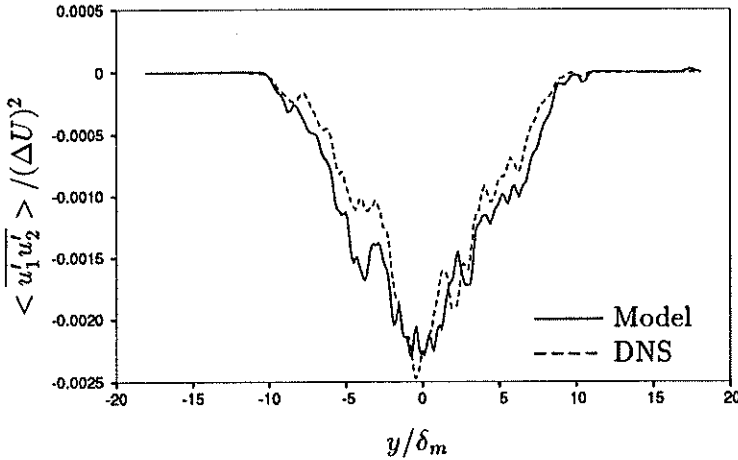


FIGURE 7. y -distribution of the SGS-Reynolds shear stress using Generalized normal stress model (Mixing layer)

2.2 Generalized SGS normal-stress model

Although it is shown that the SGS-EVM based on the ‘major-gradient stress’ model (Eq. (4) with $E = \overline{u_2'u_2'}$) shows a high correlation with the DNS data, this model does not preserve the tensoral invariance of the SGS Reynolds stresses. This drawback can be circumvented by generalizing this model as follows (Durbin 1991 and Horiuti 1993):

$$\overline{u_i'u_j'} = \delta_{ij}\left(\frac{2}{3}\overline{E_G} + \frac{2}{3}P\right) - \nu_{eii}\frac{\partial\overline{u}_j}{\partial x_l} - \nu_{ejl}\frac{\partial\overline{u}_i}{\partial x_l} \quad (7)$$

where

$$\nu_{eij} = C_\nu \frac{\Delta}{\overline{E}_G^{1/2}} \overline{u'_i u'_j}, \quad P = \nu_{eml} \frac{\partial \overline{u}_m}{\partial x_l}.$$

This model is not identical to the conventional eddy viscosity model. In this model, the velocity scale is chosen as the component normal to the principal shear plane, for example, for $\overline{u'_1 u'_2}$, the most dominant term in the RHS becomes $n u_{e22} \partial \overline{u}_1 / \partial x_2$.

The generalized normal stress model is tested in the same manner as in the last section. The results for the high- Re channel flow ($C_\nu = 0.23$) and the mixing layer ($C_\nu = 0.26$) are shown in Figs.6 and 7, respectively. It is evident that the generalized normal stress model shows a high correlation with the previous normal stress model.

It should be pointed out that the generalization of the normal stress model without losing the tensorial invariance is not unique, e.g. terms such as

$$-\nu_{eii} \frac{\partial \overline{u}_l}{\partial x_j} - \nu_{eji} \frac{\partial \overline{u}_l}{\partial x_i} \quad (8)$$

can be added to Eq. (7). When these terms are included, however, the term $\nu_{e11} \times \partial \overline{u}_1 / \partial x_2$ causes a large peak in the model for the $\overline{u'_1 u'_2}$ profile in the channel flow. For this reason, the terms in Eq. (8) were excluded when generalizing the normal stress model.

2.3 Approximation method of the SGS turbulent energy

To effectively use the model advocated in the previous sections, a model for the normal stresses is needed. We can either carry equations for the normal stresses or estimate the energy in the subgrid-scales from the energy in the large scales. In testing the scale-similarity model of Bardina (1983), Horiuti (1993) found good correlation between the model and the data. The model reads,

$$\begin{aligned} \overline{E}_G &= C_K (\overline{u}_l - \overline{\overline{u}}_l) (\overline{u}_l - \overline{\overline{u}}_l), \\ \overline{u'_2 u'_2} &= C_N (\overline{u}_2 - \overline{\overline{u}}_2) (\overline{u}_2 - \overline{\overline{u}}_2), \end{aligned} \quad (9)$$

where a constant different from unity was needed. It was pointed out that the optimized model constants C_K and C_N were not equal to unity because the scale-similarity model provides a partial estimate of the whole SGS fluctuations which resides in the vicinity of the cutoff-wave number ($= \pi/\Delta$). The poor performance of the model when these coefficients are set equal to unity can be evidenced by the fact that in this case the SGS flatness parameter A becomes identically zero (purely two-dimensional state). We have optimized C_K/C_N for the low- Re , the high- Re channel, and the mixing layer flows and found 7.0/12.0, 7.0/ 9.0, and 9.0/ 12.0 to be representative values for these flows. We note that they are slightly (but tolerably) sensitive to the type of flow field, and that they are generally close to each other. A representative comparison of the model prediction with the DNS data for $\overline{u'_1 u'_1}$ is shown for the high- Re channel flow and the mixing layer in Figs. 8 and 9, respectively. We find a good agreement with the DNS data for both flows.

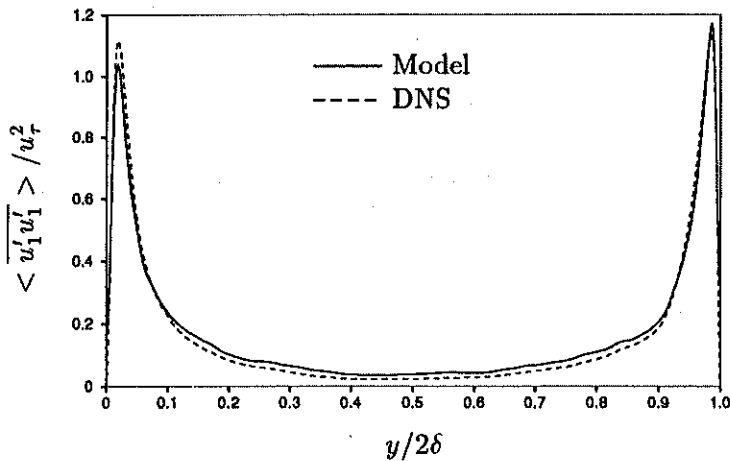


FIGURE 8. y -distribution of $\overline{u_1' u_1'} / u_\tau^2$ using the Bardina model (Channel at $Re_\tau = 790$)

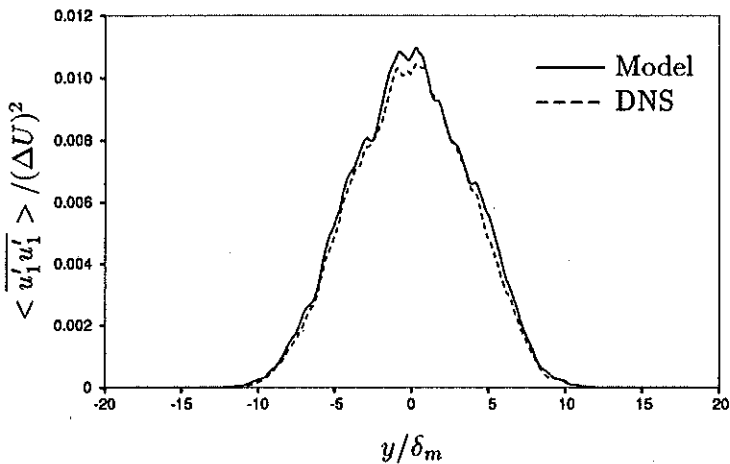


FIGURE 9. y -distribution of $\overline{u_1' u_1'} / (\Delta U)^2$ using the Bardina model (Mixing layer)

3. Future plans

An ultimate goal of the present study is to develop an SGS model which yields good predictions of turbulent flows in a complex geometry. Of particular interest is the flow over a backward-facing step. In this case, while the flow is bounded by the walls, the internal mixing layer present in this flow plays a major role in setting the turbulence levels. In this work, a proper velocity scale for the SGS-EVM viscosity was determined for the fully developed channel and the mixing layer flows. In the channel, a clear advantage over more conventional treatments was shown by using the normal stress. It was also shown that the SGS normal stress is equally

useful as the total SGS turbulent energy for modeling in mixing layers. The model constant in the normal-stress model was found to be fairly independent of the type of flow. It was demonstrated that a generalized normal stress model can be used as an alternative method for the normal-stress model, and the tensorial invariance which is violated in the normal-stress model can be recovered. This generalized normal-stress model will be tested in a backward-facing step flow in both 'a priori' and 'a posteriori' manner in the future. Although the Bardina model constants C_K and C_N in Eq. (9) are rather consistent in three different flow fields, some variance was noticed. An attempt to determine these coefficients more accurately using the Dynamic scale model approach is currently underway.

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