

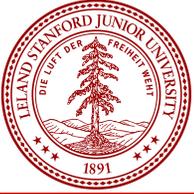


# **Reverse Engineering the GPS and Galileo Transmit Antenna Side Lobes**

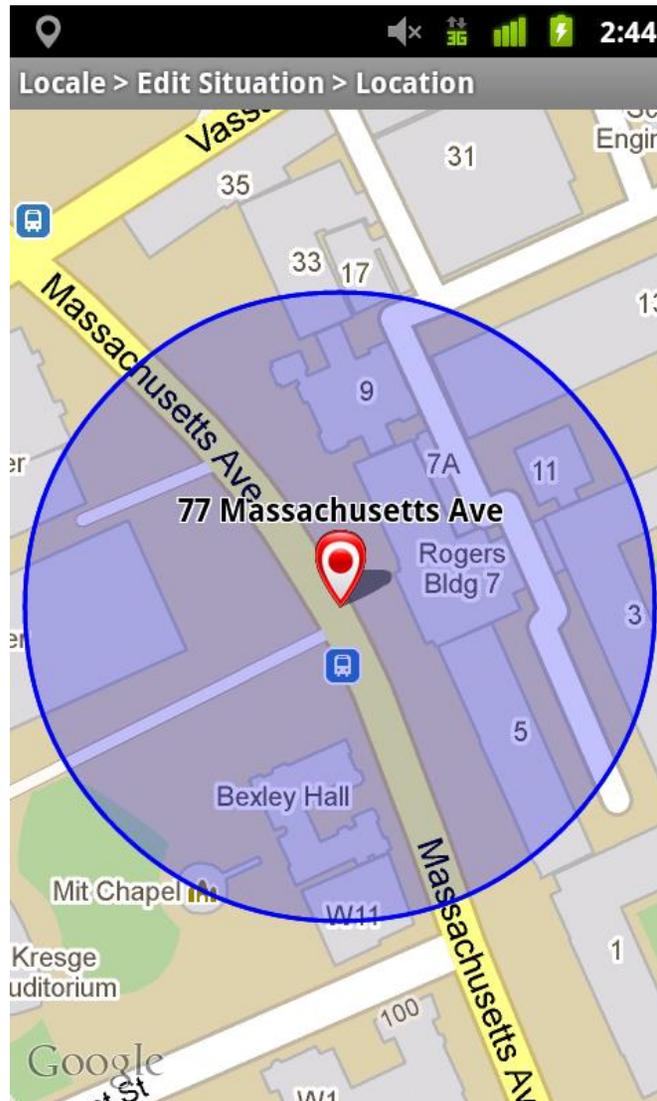
**SCPNT Symposium**  
November 11, 2015

**Shankar Ramakrishnan**

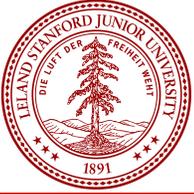
**Advisor: Per Enge**



# Location, Location, Location!



Courtesy: [www.techprone.com](http://www.techprone.com)



# Space Missions Benefiting from GNSS

## ■ Low Earth Orbit Missions (LEO)

- Attitude Determination and Control
- GNSS Radio Occultation

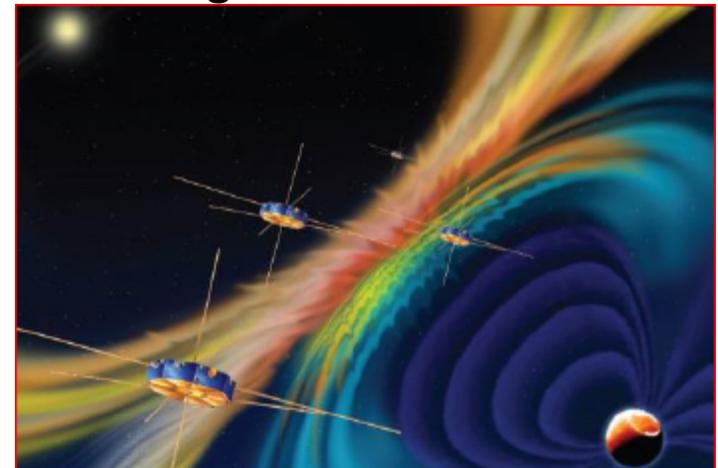


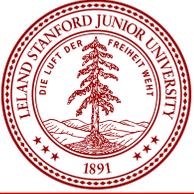
## ■ Geostationary Missions (GEO)

- Closer Spacing of Satellites in Geostationary Arc
- GEO Satellite Servicing and Decommissioning

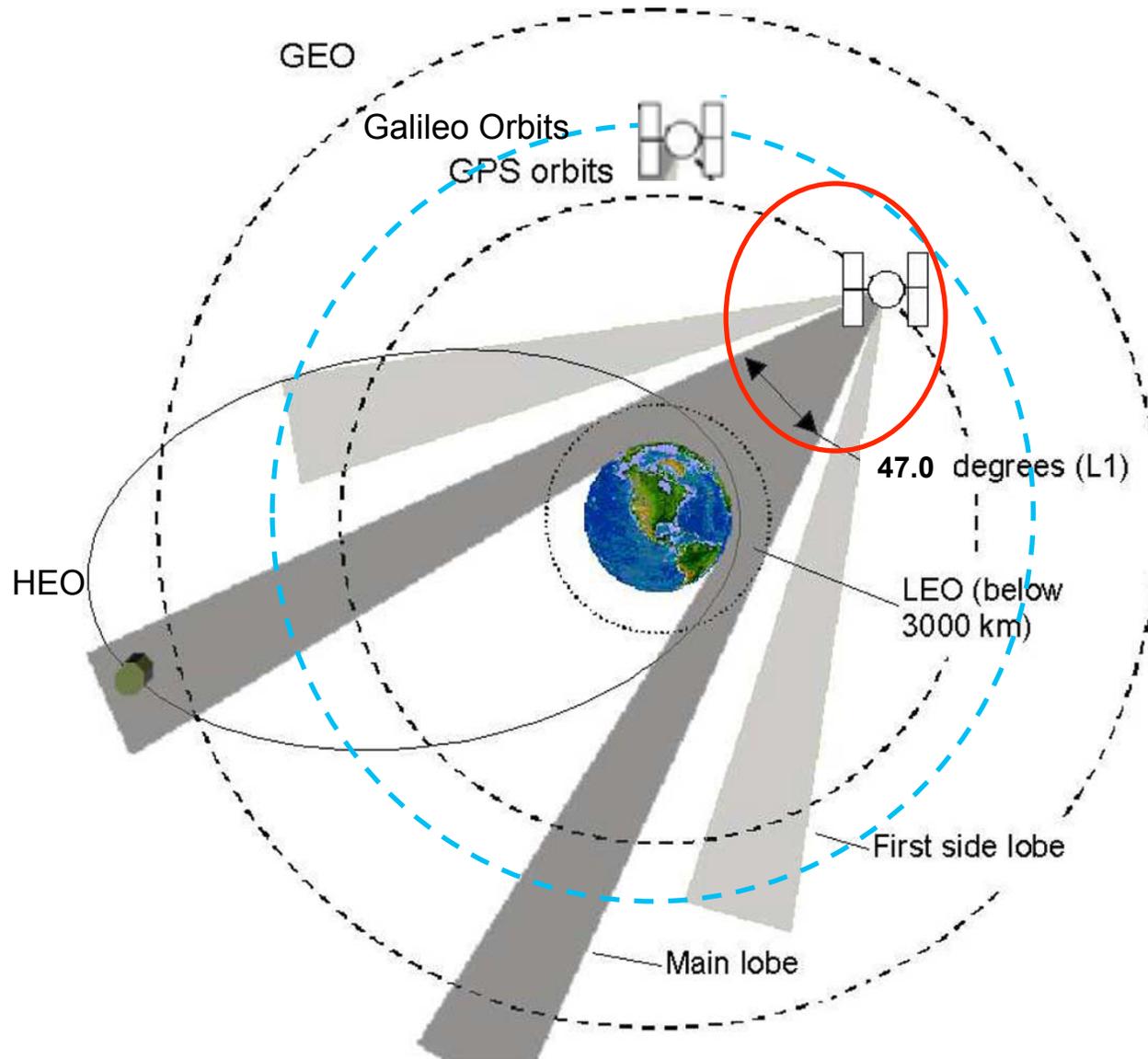
## ■ Highly Elliptical Orbit Missions (HEO)

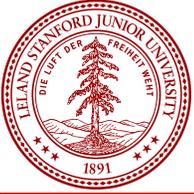
- Scientific Formation Flying/Constellation Missions
- Space Weather Observations
- En-route Lunar Navigation Support





# Using GNSS at GEO and Beyond



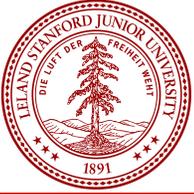


# GPS in Space – Current State of Art



## Use of GPS in the Space Service Volume (SSV)

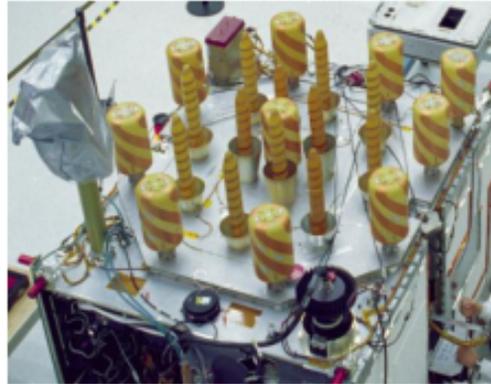
- **GPS signals in High Earth Orbit and Geosynchronous Altitude utilized by multiple DOD, NASA & NOAA programs**
  - SBIRS, ANGELS, Classified Programs
  - GOES-R, MMS
- **Autonomous navigation enables new mission needs and significantly improves PNT performance over past methods**
  - GPS Ephemeris and timing data can be provided near real time with collected satellite products
  - Achievable accuracy is greatly improved over typical methods using ground based ephemeris processing via ranging and angle measurements
- **NASA activities have included:**
  - Conducting flight experiments to characterize GPS performance in SSV
  - Development of new weak signal GPS receivers for spacecraft in Geostationary or highly elliptical orbits
  - Working with the GPS Directorate and DoD community to formally document GPS requirements for space users
  - International coordination to encourage other GNSS constellations (e.g, Galileo, GLONASS, BeiDou) to specify interoperable SSV capabilities
  - Developing missions and systems to utilize GPS signals in the SSV



# GNSS Transmit Antenna Array



GPS IIA (Credits: Geo++)

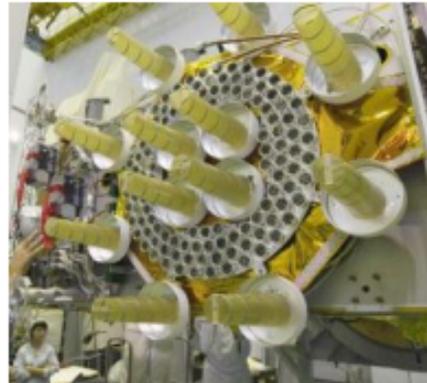


GPS IIR (Credits: Lockheed)



GPS IIF (Credits: Boeing)

**GPS and GLONASS Satellites have 12 helical antennas arranged in two concentric circles**

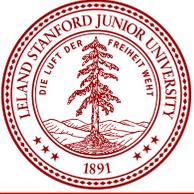


GLONASS-M (Reshetnev)



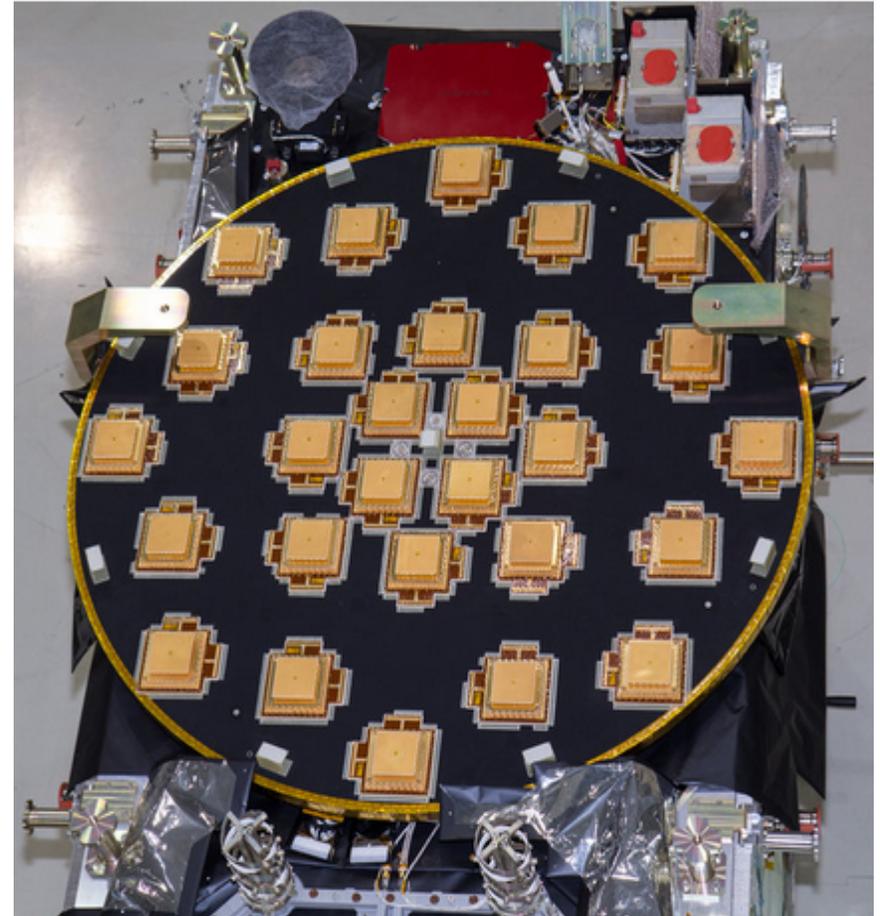
GLONASS-K1 (Reshetnev)

Czopek, F.M., S. Shollenberger (1993): Description and Performance of the GPS Block I and II L Band Antenna and Link Budget. ION GPS-93, Salt Lake City, Utah, 37-43.

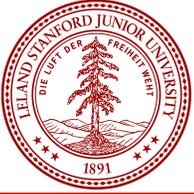


# Galileo Full Operational Capability Satellite Antenna Array Design

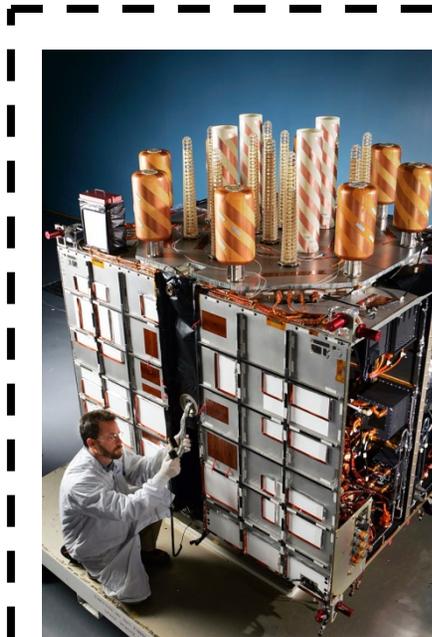
- Uses a patch antenna array design
- No publically available information regarding design parameters.
- Side lobe characteristics unknown



Courtesy: [www.technology.org](http://www.technology.org)

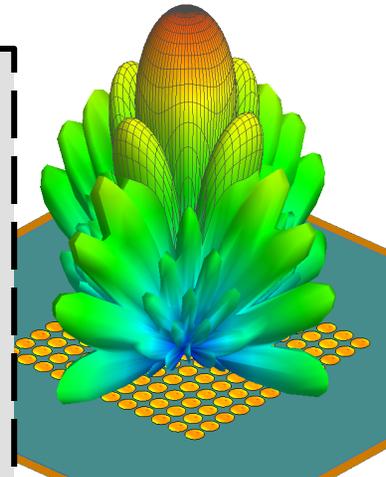


# Transmit Antenna Modeling Design Overview



- Isoflux Radiation Pattern for the Antenna Mainlobe
- Constant Circular Polarization Axial Ratio over Mainlobe
- Good Sidelobe Suppression
- Minimal Gain/Axial Ratio variation over bandwidth of interest

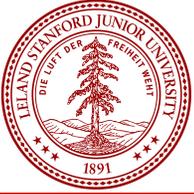
**FEKO**  
Comprehensive Electromagnetic Solutions



Establish transmit antenna specifications

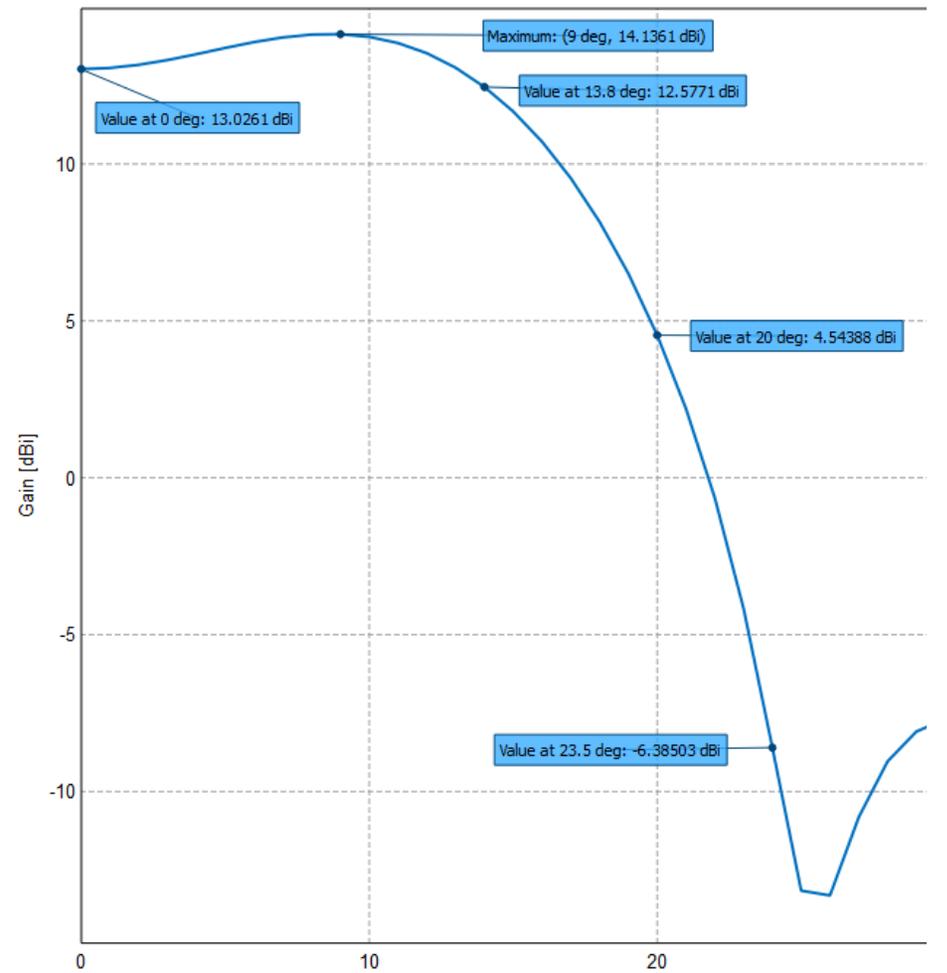
Create CAD Model of Transmit Antenna

High Fidelity Numerically analyze array design

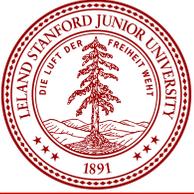


# GPS III Space Service Volume (SSV) Specifications

- **L1C signal strength at EOL:**
  - Terrestrial : -157 dBW
  - At GEO Orbit: -182.5 dBW
- **Off-Axis Relative Power shall not decrease:**
  - More than 2 dB from Edge-of-Earth to nadir
  - More than 10 dB from EOE to 20 degrees off nadir
  - More than 19.5 dB from EOE to 23.5 degrees off nadir
- **Monotonic drop off in power between EOE and  $\pm 23.5$  degrees off nadir**

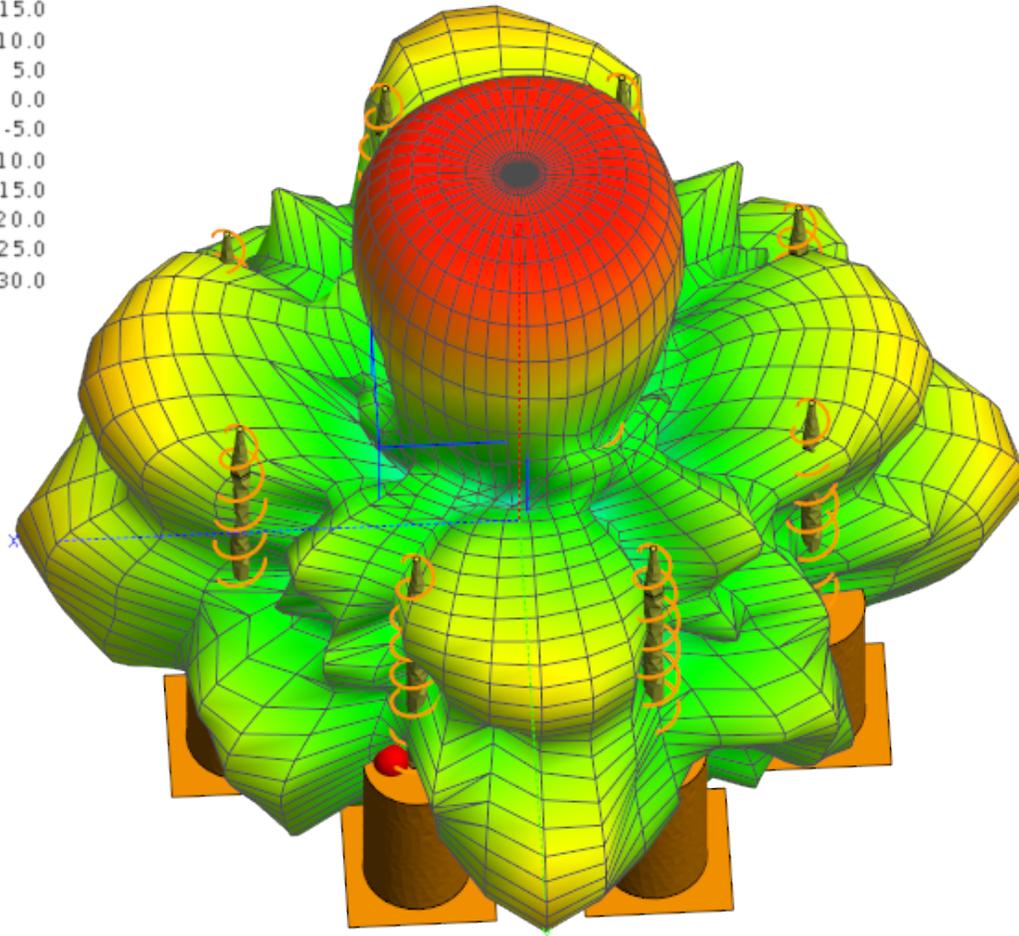


Total Gain [dBi] (Frequency = )

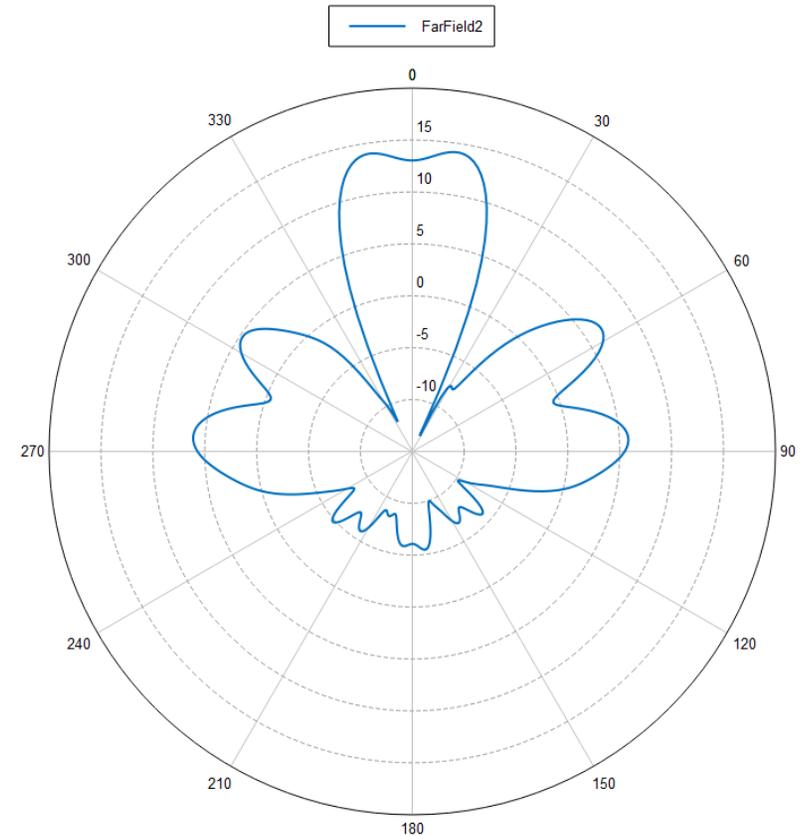


# Reverse Engineered GPS Block III Transmit Antenna 3-D Beam Pattern

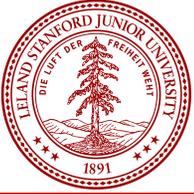
Total Gain [dBi]



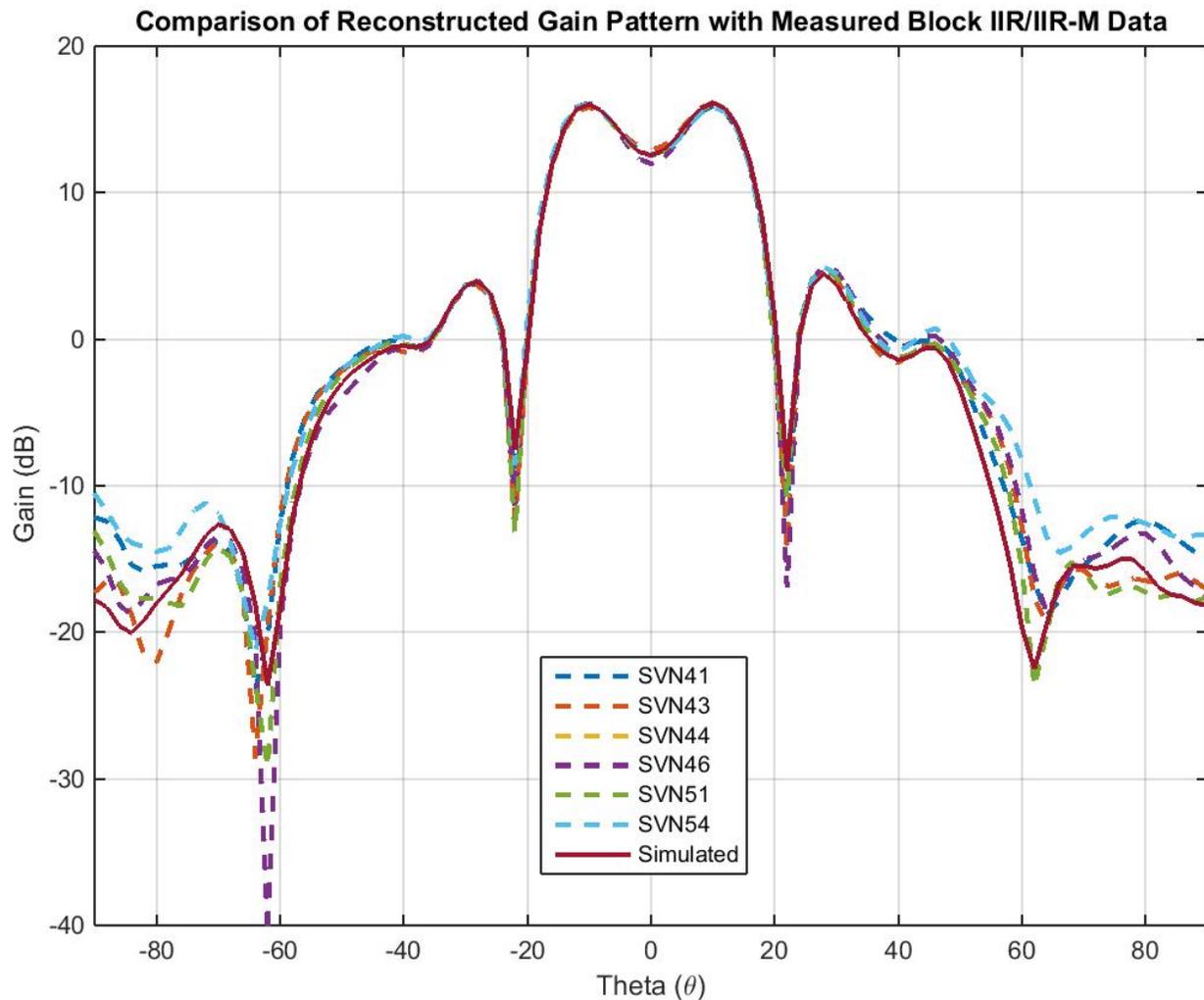
2-D Antenna Beam Pattern

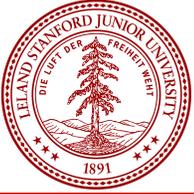


Total Gain [dBi] (Frequency = 1.57542 GHz; Phi = 0 deg)

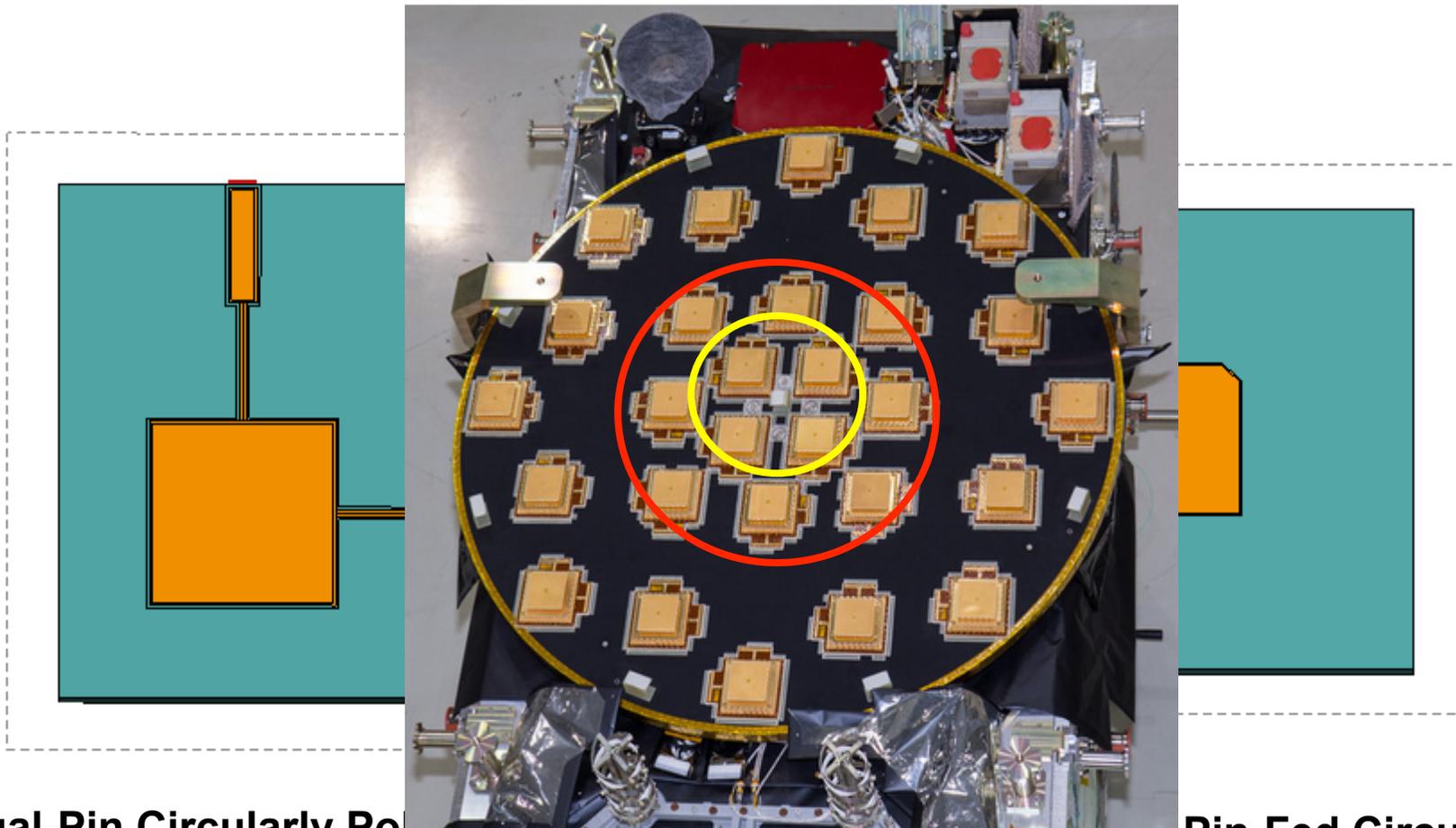


# Comparison of High-Fidelity Model versus Measured Block IIR/IIR-M Antenna Data





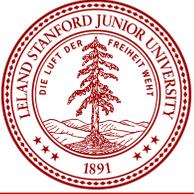
# Possible Single Element of Galileo Transmit Array



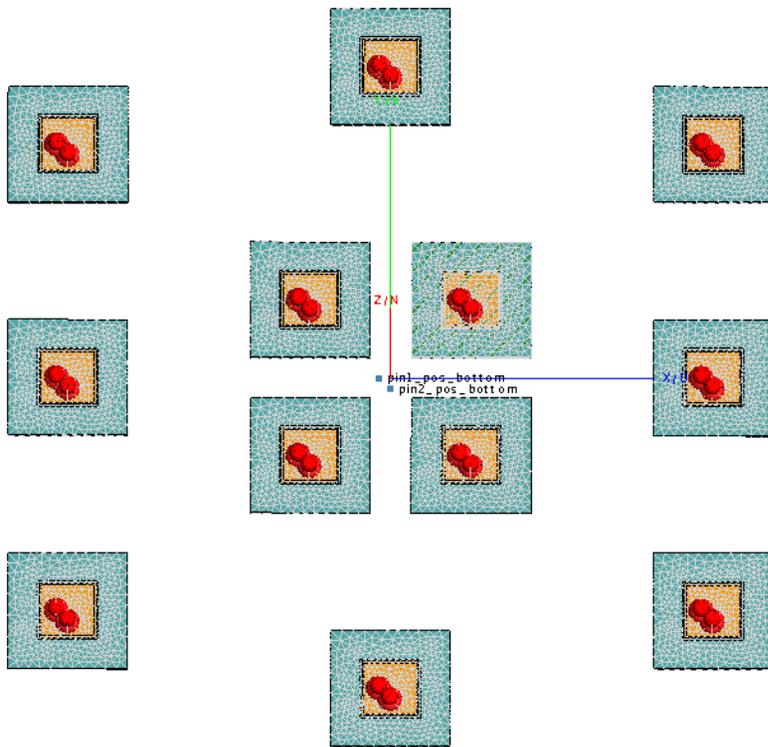
**Dual-Pin Circularly Polarized Patch**

**Square Truncated Pin-Fed Circularly Polarized Patch**

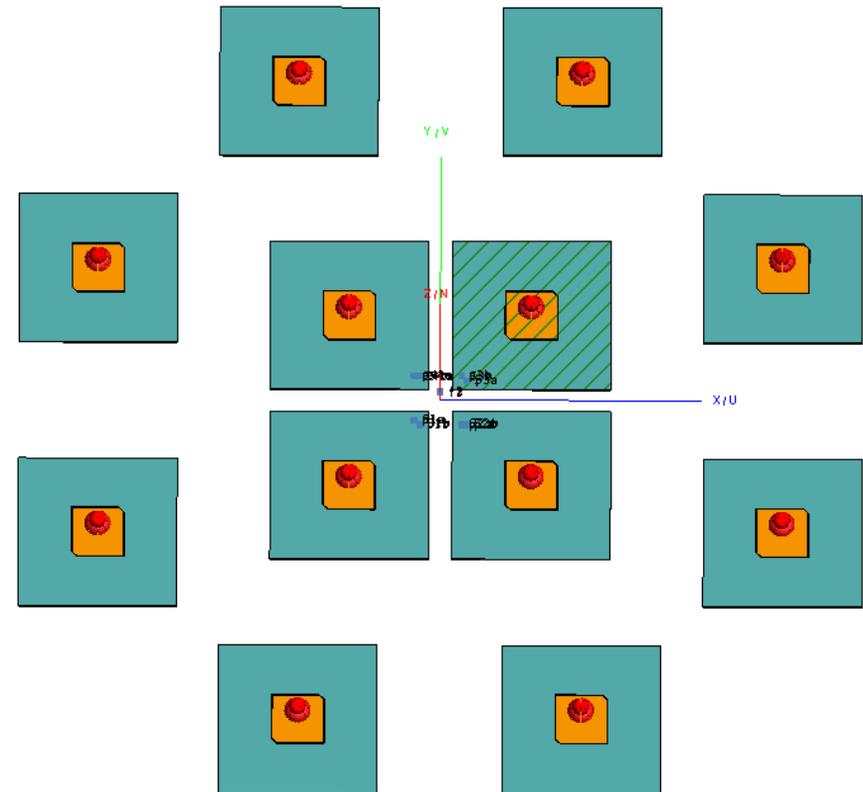
Pozar, David (Editor), *Microstrip Antennas: The Analysis and Design of Microstrip Antennas and Arrays*, Wiley-IEEE Press, 1995



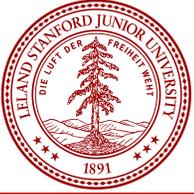
# Circularly Polarized Patch Antenna Arrays



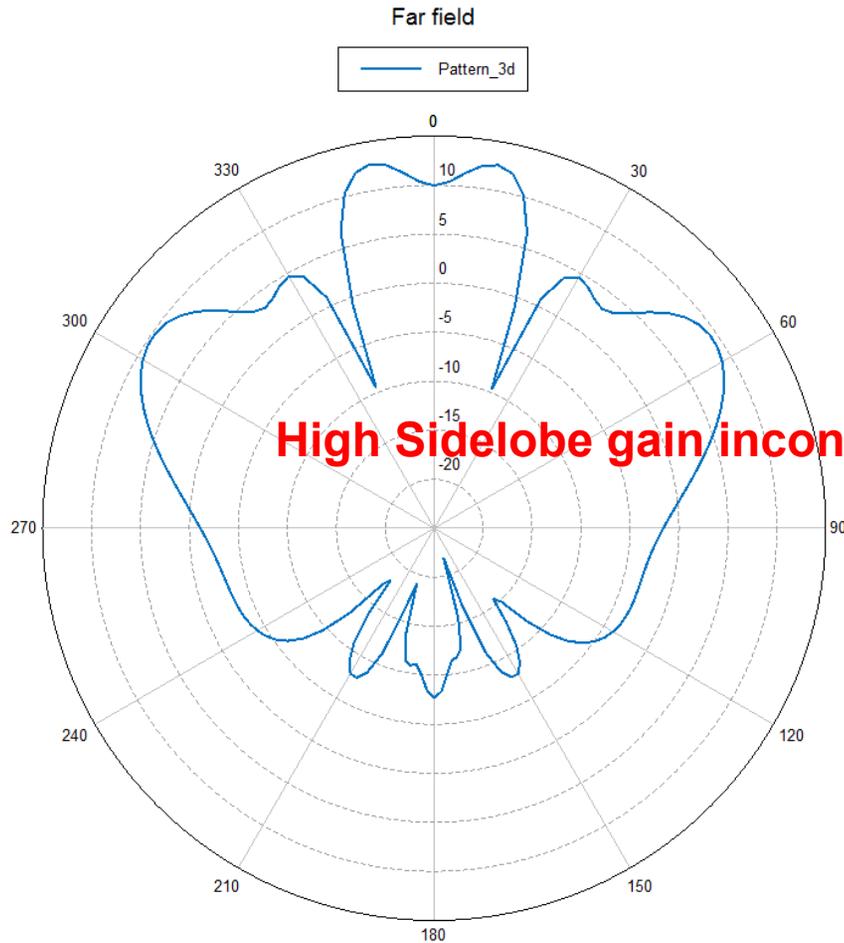
Dual-Pin Circularly Polarized Patch Array



Square Truncated Pin-Fed Circularly Polarized Patch Array

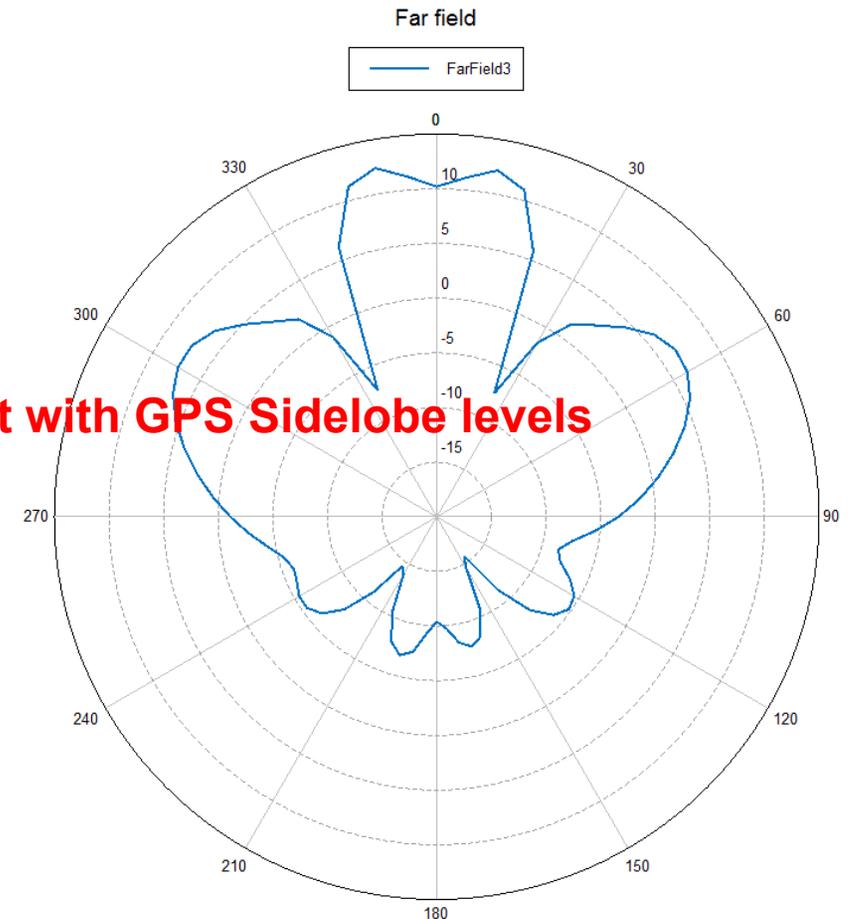


# Dual-Pin Circularly Polarized Patch Antenna Gain Pattern



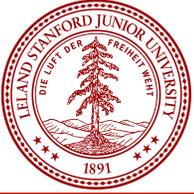
Total Gain [dBi] (Frequency = 1.57542 GHz; Phi = 0 deg) - PEC\_Rect\_Dual\_Pin\_Model\_Array\_Outer\_Dist\_3

## Dual-Pin Circularly Polarized Patch

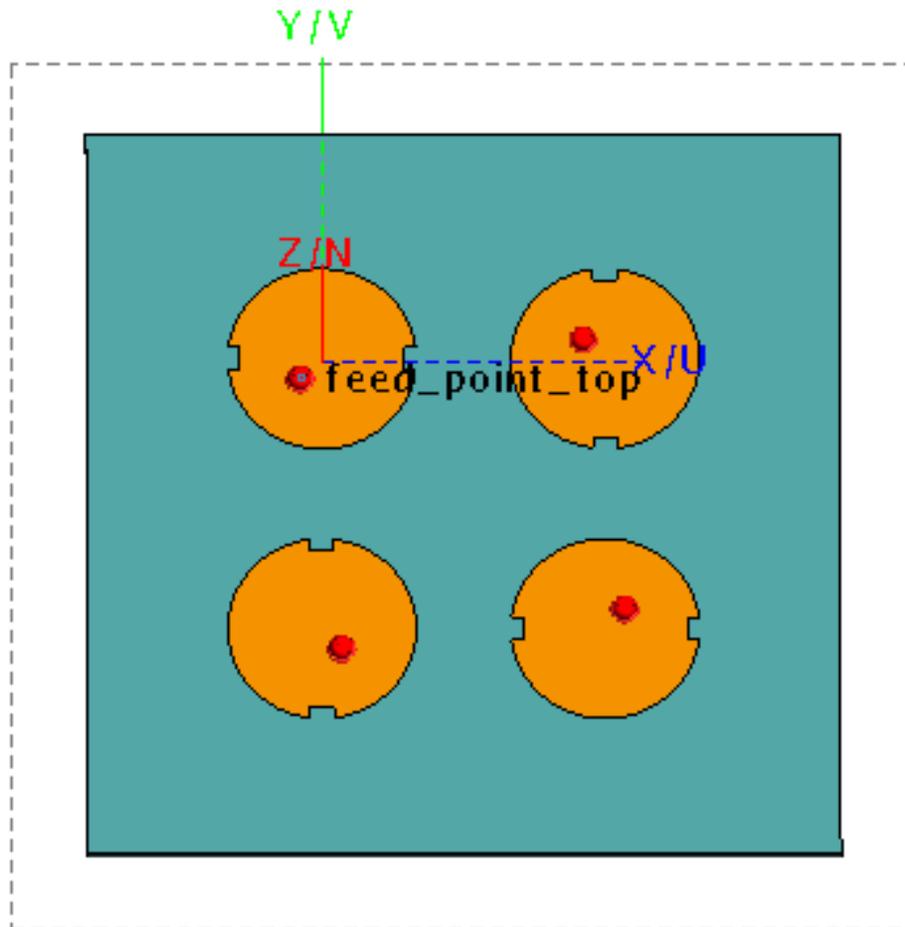


## Square Truncated Pin-Fed Circularly Polarized Patch

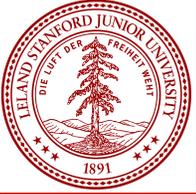
Pozar, David (Editor), *Microstrip Antennas: The Analysis and Design of Microstrip Antennas and Arrays*, Wiley-IEEE Press, 1995



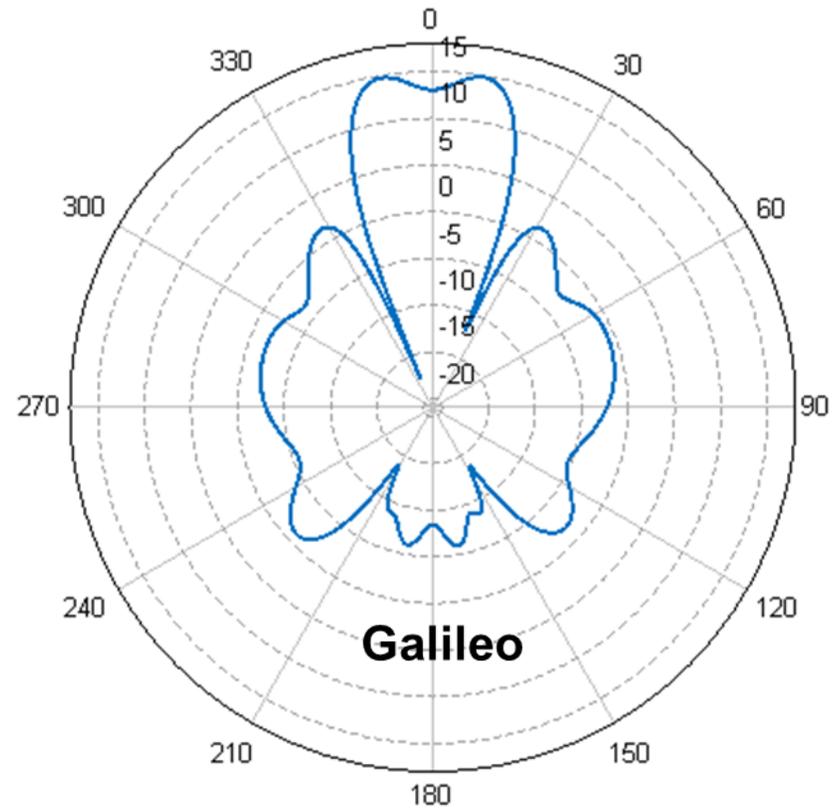
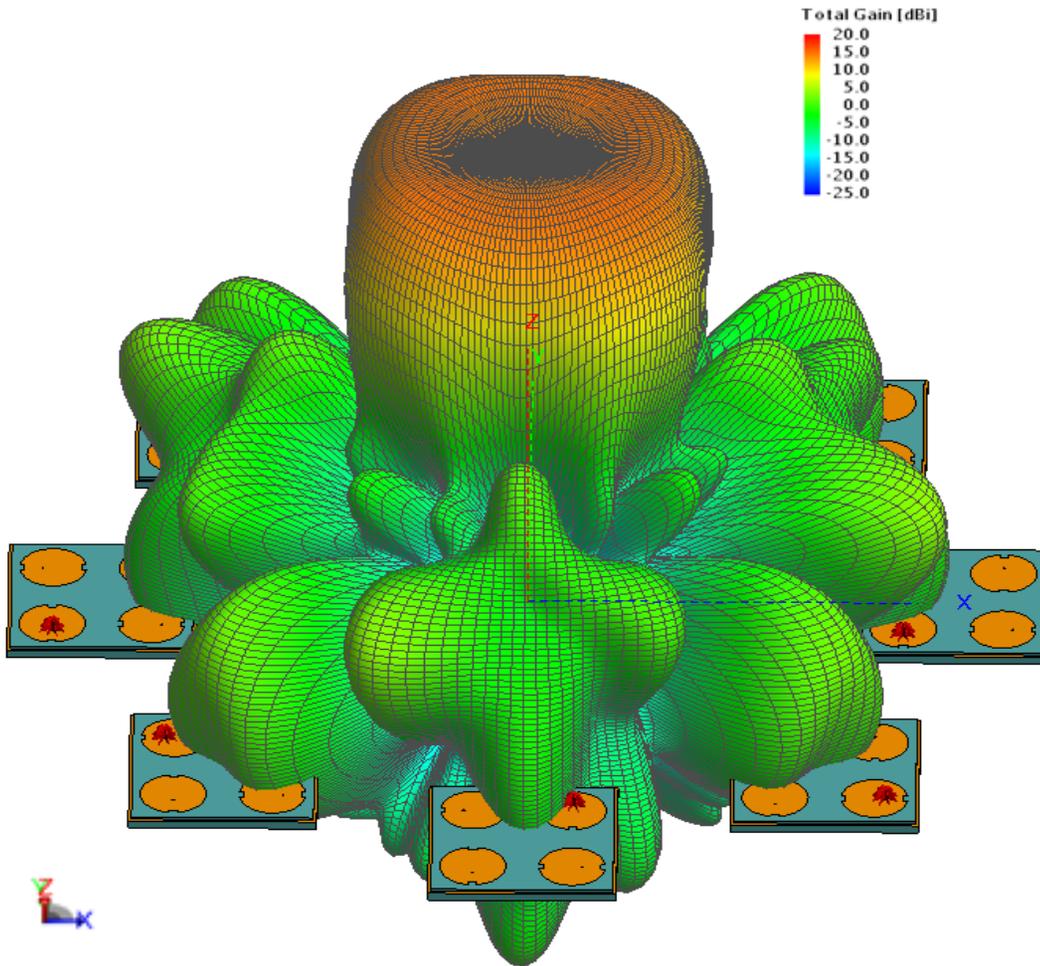
# 2x2 Sequentially Rotated Patch Antenna Array



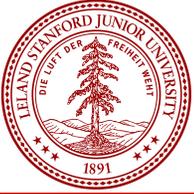
- Isoflux Radiation Pattern for the Antenna Mainlobe
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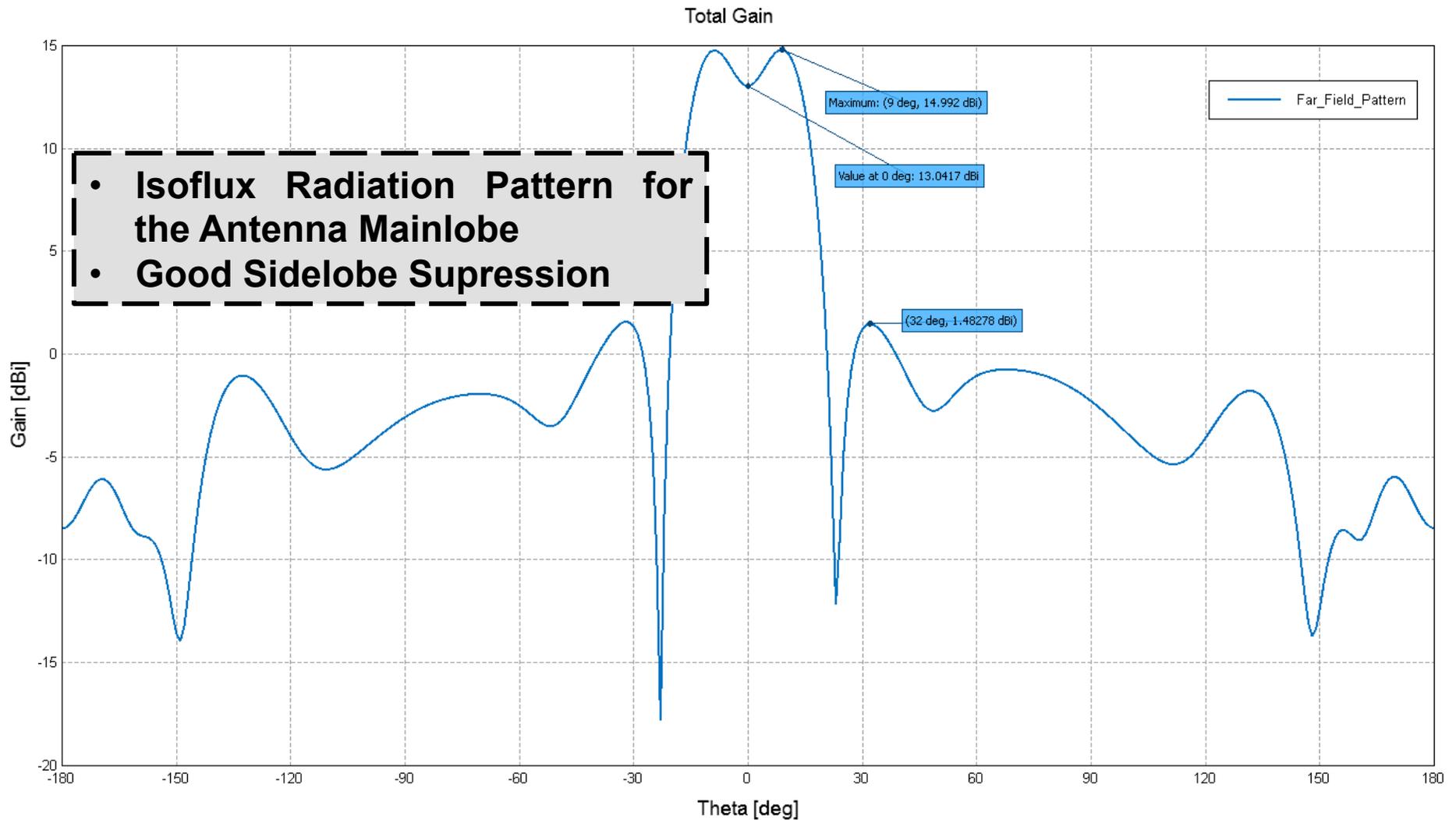
# Galileo Transmit Antenna Gain Profile



Total Gain [dBi] (Frequency = 1.57542 GHz; Phi = 0 deg)



# Sequentially Rotated Patch Antenna Array Gain Profile

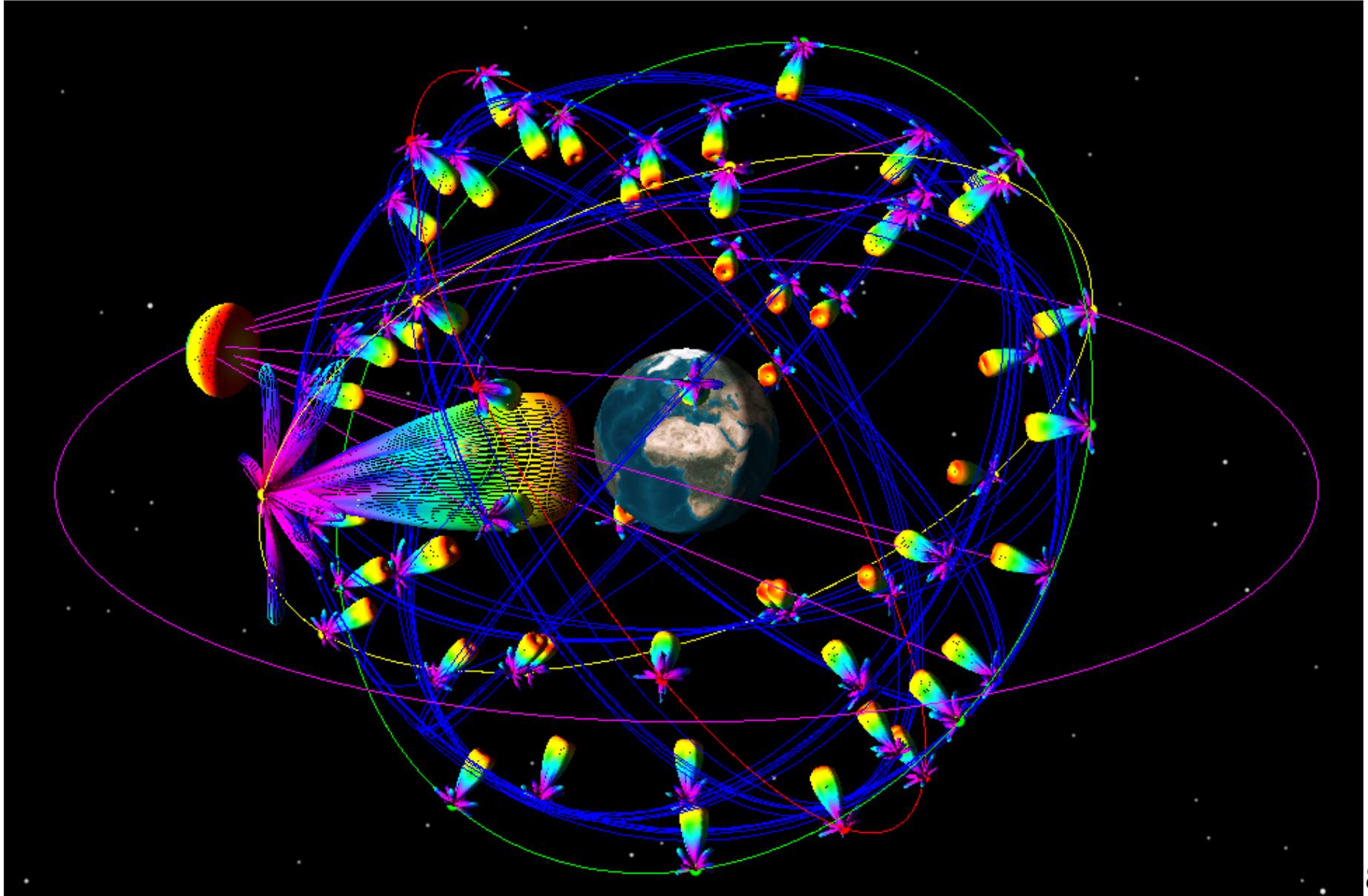


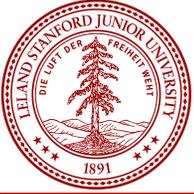
Total Gain [dBi] (Frequency = 1.57542 GHz; Phi = 0 deg)



# GNSS Satellite Visibility at GEO

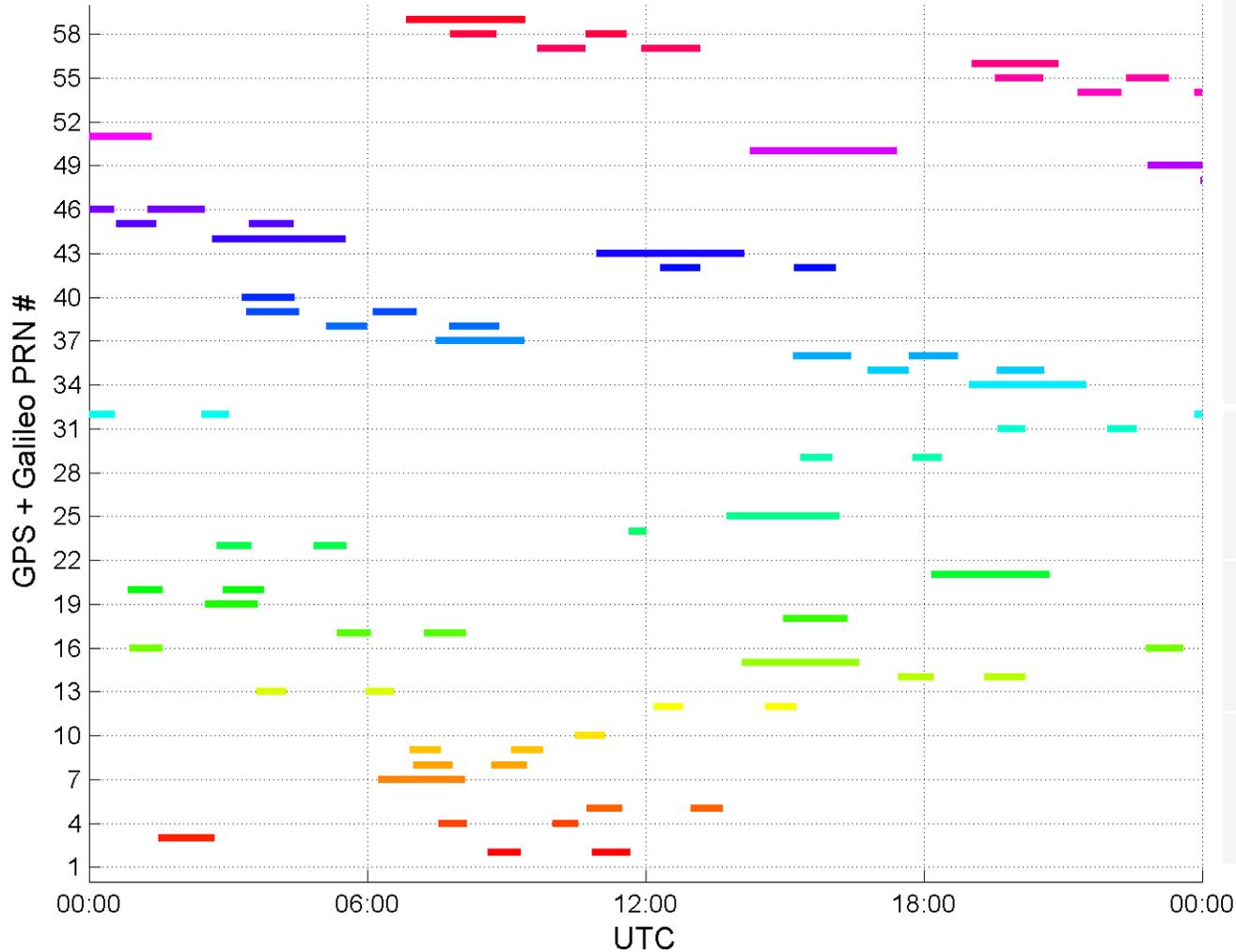
## Main & Side Lobes



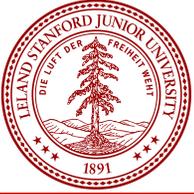


# GPS + Galileo Visibility at GEO Main Lobe Only

GPS + Galileo Visibility aboard ANIK F1R  
Received Signal Threshold: -182.5 dBW



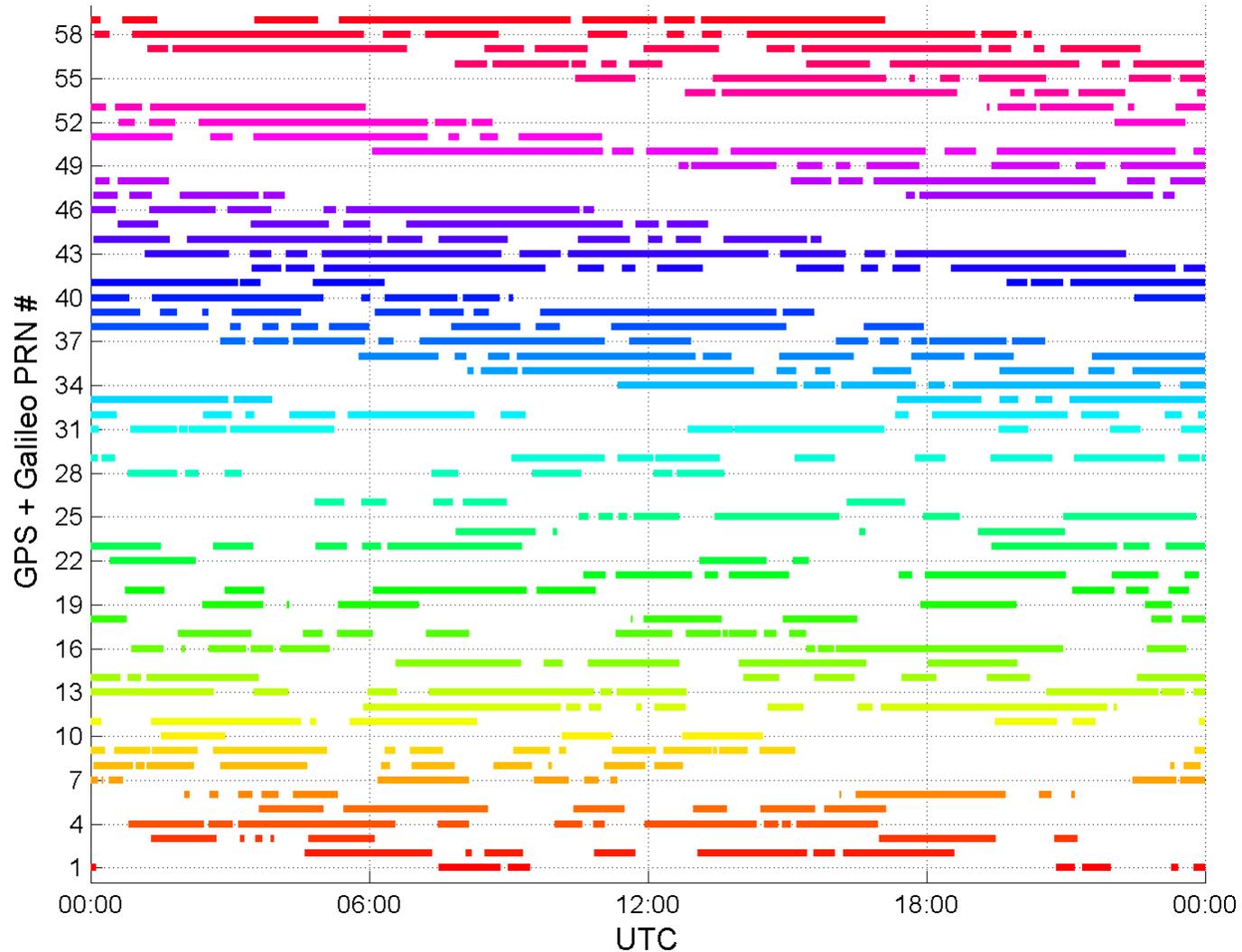
Constellation	Visibility over 24 hours	
	At least 1 SV Visible	At least 4 SVs visible
32 SV GPS Only	80.02 %	6.89 %
27 SV Galileo Only	94.51 %	12.19 %
59 SV GPS + Galileo	100 %	40.82 %

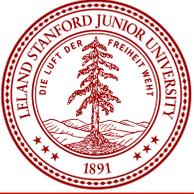


# GPS + Galileo Visibility at GEO

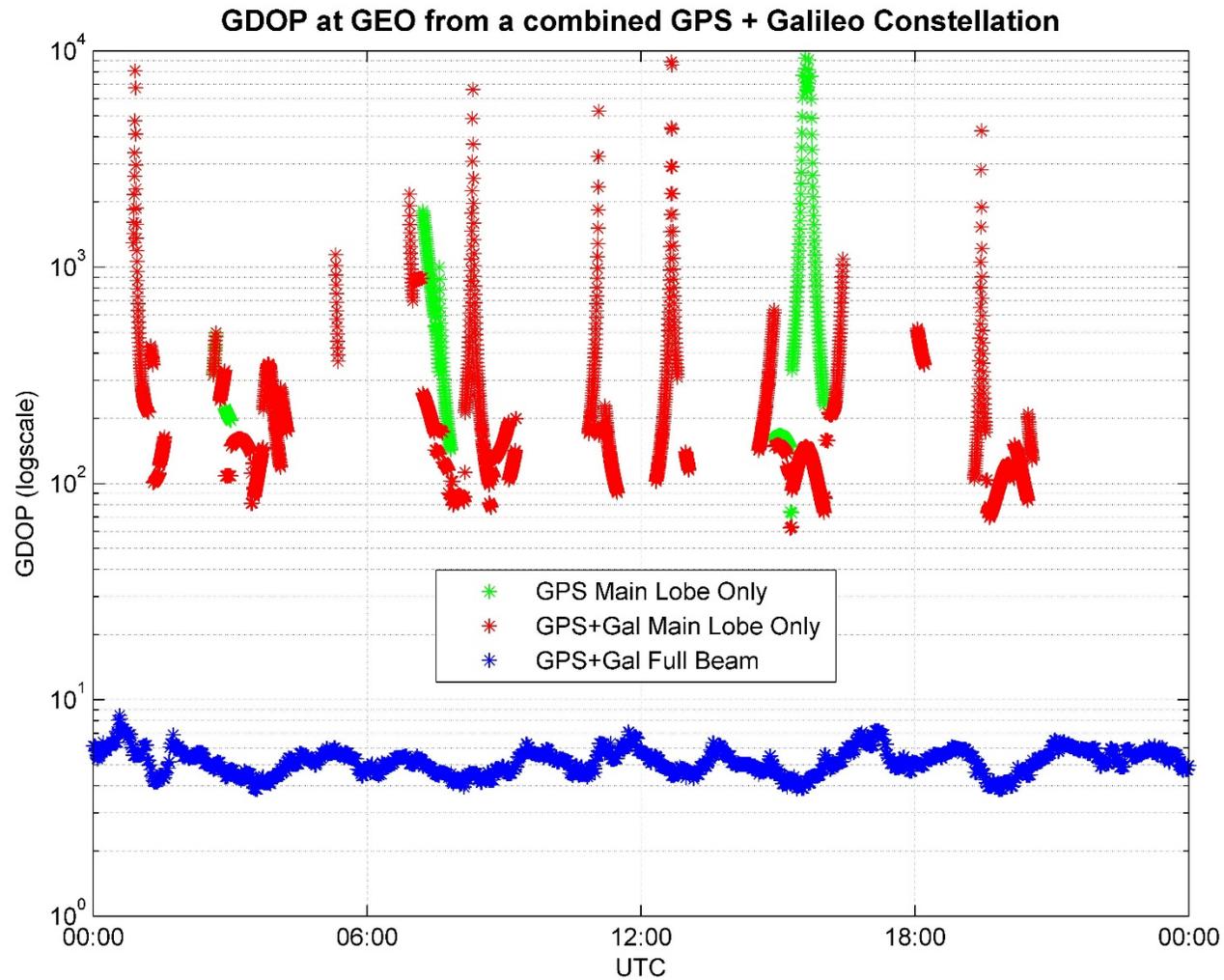
## Main & Side Lobes

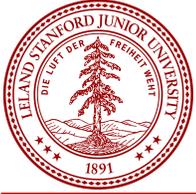
GPS + Galileo Visibility aboard ANIK F1R  
Received Signal Threshold:  $-182.5$  dBW





# GPS + Galileo Combined Constellation GDOP





## Conclusion

- **First to successfully “reverse engineered” expected GPS and Galileo transmit antenna gain patterns**
  - Both main lobe and side lobe performance was quantified
- **Evaluated satellite availability and expected performance at GEO using the main lobe only and main + side lobes**
- **Significant enhancement in availability and accuracy from a combined GPS + Galileo constellation enabling autonomous navigation at GEO**
  - Common L1C signal allows for single receiver implementation capable of tracking both constellations